

On the Advantages of Being Periodic

rEvolving Horizons in Systems and Control

*In Honor and Appreciation of
Elmer Gilbert*

Evolving Horizons in Systems and Control





Goal



- 33 years of perspective on what my PhD research was about!

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- Marquez—*One Hundred Years of Solitude*
 - “Just like Aureliano,” Ursula exclaimed. “It’s as if the world were repeating itself.”

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- Ecclesiastes (Koheleth)
 - The sun rises and the sun sets,
and hurries back to where it rises.
 - The wind blows to the south
and turns to the north;
round and round it goes,
ever returning on its course.
 - All streams flow into the sea,
yet the sea is never full.
To the place the streams come from,
there they return again.
 - What has been will be again,
what has been done will be done again;
there is nothing new under the sun.





Periodic Control



- What is it?
- Why do it?
- Why need it?

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- How to best control a system for long-term operation?
 - Ignore transients/startup
 - E.g., ascent, descent
 - Operate sustainably
 - E.g., Cruise
 - Maximize endurance
 - Minimize fuel usage rate
- Constant operation---obvious approach
- Periodic operation---why?

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Optimal Periodic Control



SIAM J. CONTROL AND OPTIMIZATION
Vol. 15, No. 5, August 1977

OPTIMAL PERIODIC CONTROL: A GENERAL THEORY OF NECESSARY CONDITIONS*

ELMER G. GILBERT†

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Optimal periodic control problem (OPC). Find $u(\cdot)$, $x(\cdot)$ and τ which minimize J subject to

$$(2.1-1) \quad J = g_0(y, x(0)),$$

$$(2.1-2) \quad g_i(y, x(0)) \leq 0, \quad i = -j, \dots, -1,$$

$$(2.1-3) \quad g_i(y, x(0)) = 0, \quad i = 1, \dots, k,$$

Gilbert, SICOPT,
1977

$$(2.1-4) \quad y = \frac{1}{\tau} \int_0^\tau \tilde{f}(x(t), u(t)) dt \in Y,$$

$$(2.1-5) \quad \dot{x}(t) = f(x(t), u(t)) \quad \text{almost all } t \in [0, T], \quad x(0) = x(\tau),$$

$$(2.1-6) \quad u(\cdot) \in \mathcal{U} = \{u(\cdot) : u(\cdot) \text{ measurable and essentially bounded on } [0, T], u(t) \in U \text{ for all } t \in [0, T]\},$$

$$(2.1-7) \quad x(\cdot) \in \mathcal{X} = \{x(\cdot) : x(\cdot) \text{ absolutely continuous on } [0, T], x(t) \in X \text{ for all } t \in [0, T]\},$$

$$(2.1-8) \quad \tau \in (0, T].$$



Optimal steady-state problem (OSS). Find u and x which minimize J subject to

$$(2.2-1) \quad J = g_0(y, x),$$

$$(2.2-2) \quad g_i(y, x) \leq 0, \quad i = -j, \dots, -1,$$

$$(2.2-3) \quad g_i(y, x) = 0, \quad i = 1, \dots, k,$$

$$(2.2-4) \quad y = \tilde{f}(x, u) \in Y,$$

$$(2.2-5) \quad f(x, u) = 0,$$

$$(2.2-6) \quad u \in U,$$

$$(2.2-7) \quad x \in X.$$



$$(2.3) \quad \mathcal{S}(\text{OPC}) = \{(u(\cdot), x(\cdot), \tau) : (u(\cdot), x(\cdot), \tau) \text{ solves OPC}\},$$

$$(2.4) \quad \mathcal{S}(\text{SS}) = \{(u(\cdot), x(\cdot), \tau) : (2.1-2)-(2,1-8) \text{ are satisfied and } u(\cdot) \text{ and } x(\cdot) \text{ are constant}\},$$

$$(2.5) \quad \mathcal{S}(\text{SSOPC}) = \mathcal{S}(\text{OPC}) \cap \mathcal{S}(\text{SS}),$$

$$(2.6) \quad \mathcal{S}(\text{OSS}) = \{(u(\cdot), x(\cdot), \tau) : (u(\cdot), x(\cdot), \tau) \in \mathcal{S}(\text{SS}) \text{ and } (u(0), x(0)) \text{ solves OSS}\}.$$

Of course, $\mathcal{S} = \emptyset$, the null set, is possible in any of the four cases. The particular circumstance $\mathcal{S}(\text{SSOPC}) = \emptyset$, $\mathcal{S}(\text{OSS}) \neq \emptyset$ implies that there exist time-dependent controls which do better than the best steady-state controls. If $\mathcal{S}(\text{SSOPC}) \neq \emptyset$ any $\psi \in \mathcal{S}(\text{SSOPC})$ is also in $\mathcal{S}(\text{OSS})$ since ψ is optimum with respect to choices in $\mathcal{U} \times \mathcal{X} \times (0, T]$ and $\mathcal{S}(\text{SS}) \subset \mathcal{U} \times \mathcal{X} \times (0, T]$. Also, it is clear that all elements of $\mathcal{S}(\text{OSS})$ and $\mathcal{S}(\text{SSOPC})$ yield identical costs J . This leads to the following.

Remark 2.1. There are three mutually exclusive possibilities:

- (i) $\mathcal{S}(\text{SSOPC}) = \mathcal{S}(\text{OSS}) \neq \emptyset$;
- (ii) $\mathcal{S}(\text{SSOPC}) = \emptyset$, $\mathcal{S}(\text{OSS}) \neq \emptyset$;
- (iii) $\mathcal{S}(\text{SSOPC}) = \mathcal{S}(\text{OSS}) = \emptyset$.

Possibility (iii) is not apt to occur since for well posed problems it is likely that $\mathcal{S}(\text{OSS}) \neq \emptyset$. Possibility (i) implies that OPC has a steady-state solution and consequently, there is no advantage (even though OPC may also have time-dependent solutions) in using time-dependent control. Possibility (ii) implies time-dependent control can do better than steady-state control (a statement which holds true even if $\mathcal{S}(\text{OPC}) = \emptyset$). Because of the importance of possibilities (i) and (ii) the following definitions are introduced.



DEFINITION 2.1. If $\mathcal{S}(\text{SSOPC}) = \mathcal{S}(\text{OSS}) \neq \emptyset$ the problem OPC is called *steady-state*.

DEFINITION 2.2. If $\mathcal{S}(\text{SSOPC}) = \emptyset$, $\mathcal{S}(\text{OSS}) \neq \emptyset$ the problem OPC is called *proper* (compare [5]).

The study of relative minima of OPC and OSS will prove to be of value, particularly in the case of steady-state minima.

DEFINITION 2.3. $(u(\cdot), x(\cdot), \tau) \in \mathcal{S}(\text{SS})$ is a *strong* {weak} *relative minimum* of OPC if there exists an $\varepsilon > 0$ such that for all $(\hat{u}(\cdot), \hat{x}(\cdot), \hat{\tau})$ which satisfy (2.1-2)–(2.1-8) and $\|\hat{x}(t) - x(0)\| < \varepsilon$ $\{\|\hat{x}(t) - x(0)\| < \varepsilon, \|\hat{u}(t) - u(0)\| < \varepsilon\}$, $t \in [0, T]$, it follows that $g_0(y, x(0)) \leq g_0(\hat{y}, \hat{x}(0))$.

DEFINITION 2.4. $(u(\cdot), x(\cdot), \tau) \in \mathcal{S}(\text{SS})$ is a *strong* {weak} *relative minimum* of OSS if there exists an $\varepsilon > 0$ such that for all (\hat{u}, \hat{x}) which satisfy (2.2-2)–(2.2-7) and $\|\hat{x} - x(0)\| < \varepsilon$ $\{\|\hat{x} - x(0)\| < \varepsilon, \|\hat{u} - u(0)\| < \varepsilon\}$ it follows that $g_0(y, x(0)) \leq g_0(\hat{y}, \hat{x})$.

In these definitions $\|\cdot\|$ denotes any norm on R^n or R^l and $\hat{y} = y$ for $u = \hat{u}$, $x = \hat{x}$, $\tau = \hat{\tau}$. Corresponding to each of the four types of relative minima, notations for the set of minima are adopted:

$$\mathcal{A}(\text{SRMSSOPC}), \mathcal{S}(\text{WRMSSOPC}), \mathcal{S}(\text{SRMOSS}), \mathcal{S}(\text{WRMOSS}).$$

For example,

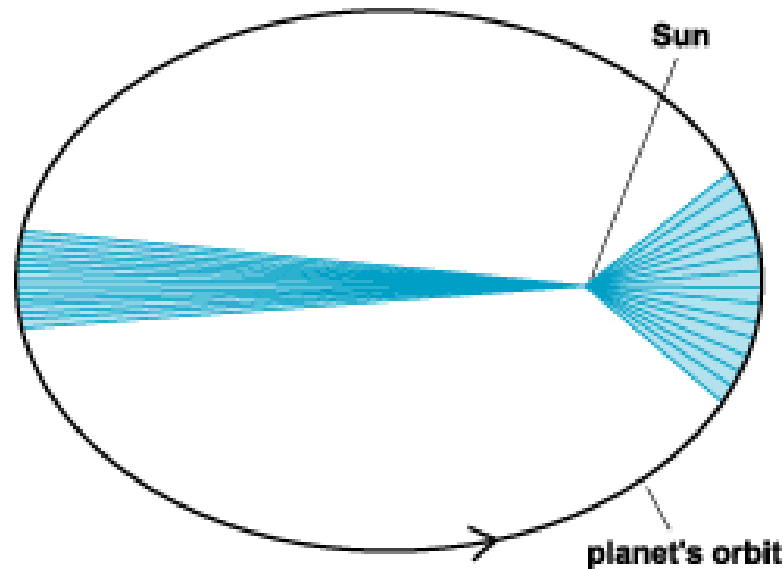
$$(2.7) \quad \mathcal{S}(\text{SRMSSOPC}) = \{(u(\cdot), x(\cdot), \tau) : (u(\cdot), x(\cdot), \tau) \in \mathcal{S}(\text{SS}) \text{ is a strong relative minimum of OPC}\}.$$

Obviously, $\mathcal{S}(\text{SSOPC}) \subset \mathcal{S}(\text{SRMSSOPC}) \subset \mathcal{S}(\text{WRMSSOPC})$ and $\mathcal{S}(\text{OSS}) \subset \mathcal{S}(\text{SRMOSS}) \subset \mathcal{S}(\text{WRMOSS})$. By using the same reasoning which led to Remark 2.1 it is easy to see that $\mathcal{S}(\text{SRMSSOPC}) \subset \mathcal{S}(\text{SRMOSS})$. However, $\mathcal{S}(\text{SRMSSOPC}) \neq \emptyset$ does not imply $\mathcal{S}(\text{SRMSSOPC}) = \mathcal{S}(\text{SRMOSS})$ because elements of $\mathcal{S}(\text{SRMSSOPC})$ do not necessarily have the same cost as elements of $\mathcal{S}(\text{SRMOSS})$. Similar reasoning applies to the case of weak relative minima. All of this is summarized in

Remark 2.2. The following conclusions are valid: $\mathcal{S}(\text{SSOPC}) \subset \mathcal{S}(\text{SRMSSOPC}) \subset \mathcal{S}(\text{WRMSSOPC})$, $\mathcal{S}(\text{OSS}) \subset \mathcal{S}(\text{SRMOSS}) \subset \mathcal{S}(\text{WRMOSS})$, $\mathcal{S}(\text{SSOPC}) \subset \mathcal{S}(\text{OSS})$, $\mathcal{S}(\text{SRMSSOPC}) \subset \mathcal{S}(\text{SRMOSS})$, $\mathcal{S}(\text{WRMSSOPC}) \subset \mathcal{S}(\text{WRMOSS})$.



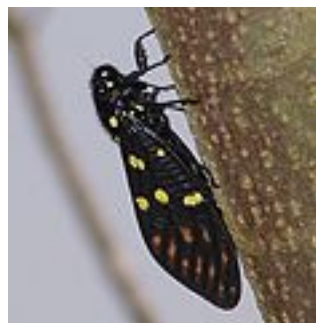
- Kepler's laws and elliptical orbits



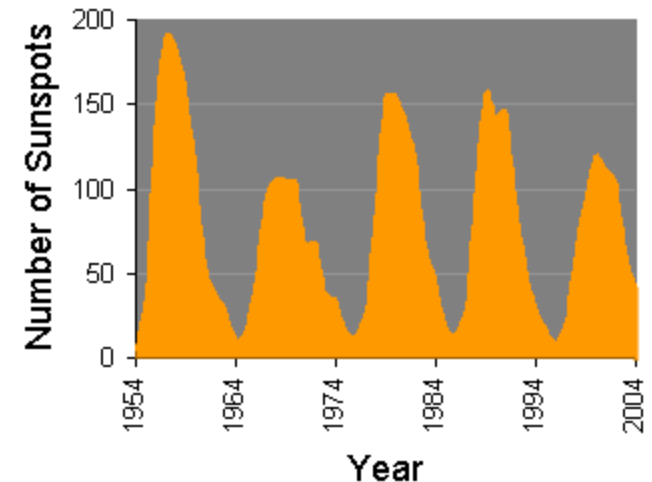
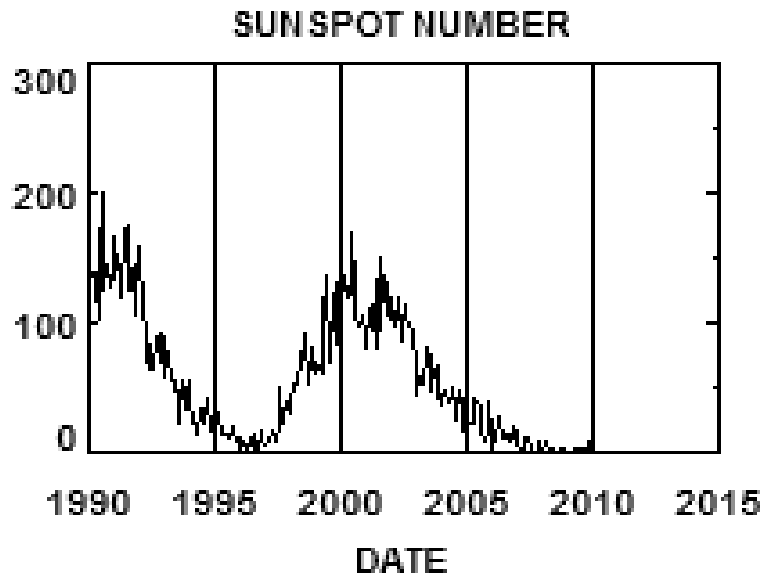
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- Cicada
 - 13 and 17 year cycles
 - Predator resistance



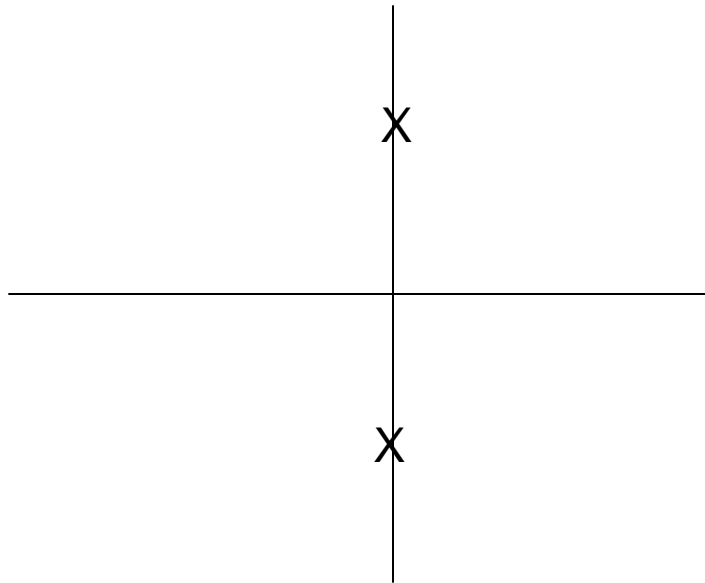
- 9-14 years, 11 years on average



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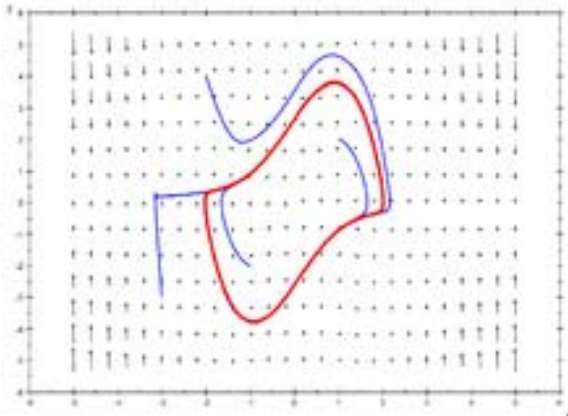
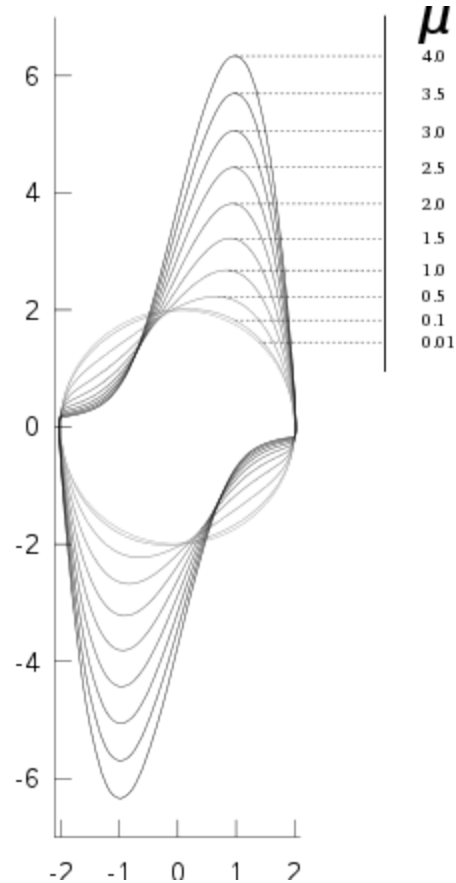
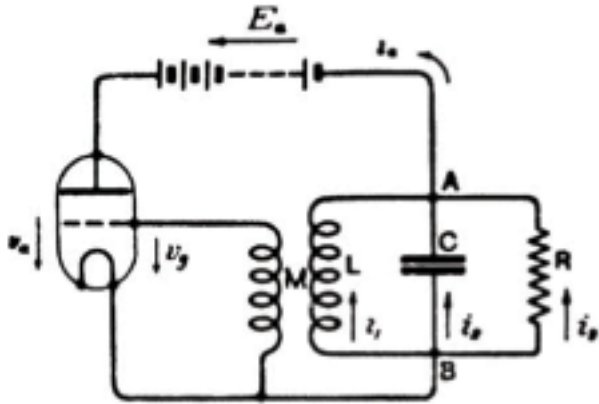


- Imaginary poles give periodic response



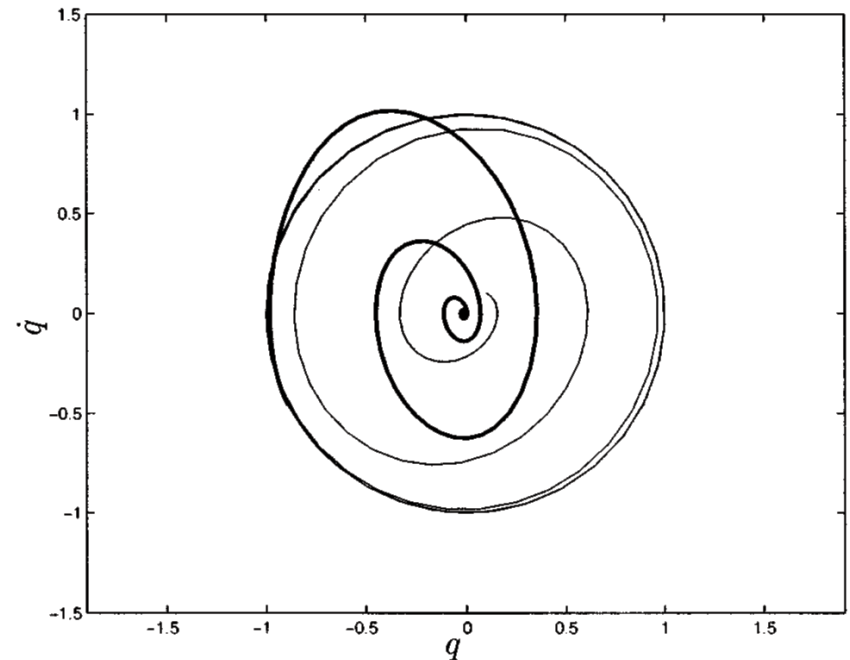
(Van der Pol 1920)

$$\ddot{x} - \mu(1 - x^2)\dot{x} + x = 0$$



2001

- To obtain a circular/sinusoidal limit cycle with amplitude a and frequency ω_n
- Speed of convergence to limit cycle determined by λ



$$\ddot{q} + \lambda(q^2 + \omega_n^{-2}\dot{q}^2 - a^2) + \omega_n^2 q = 0$$

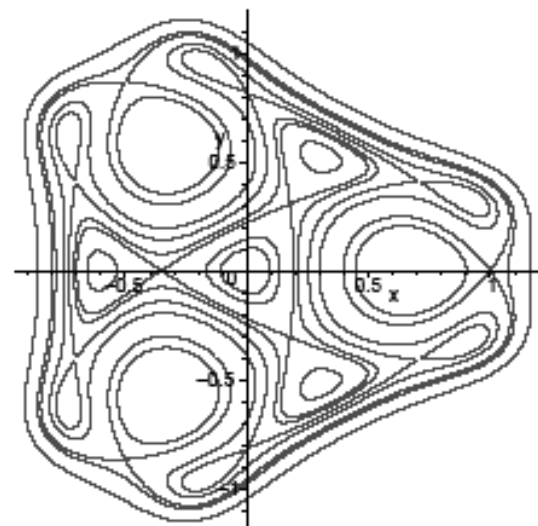


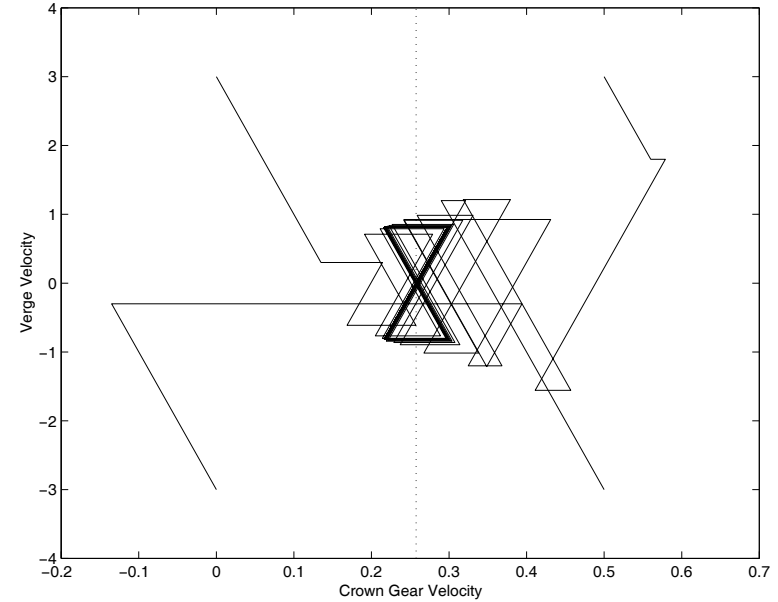
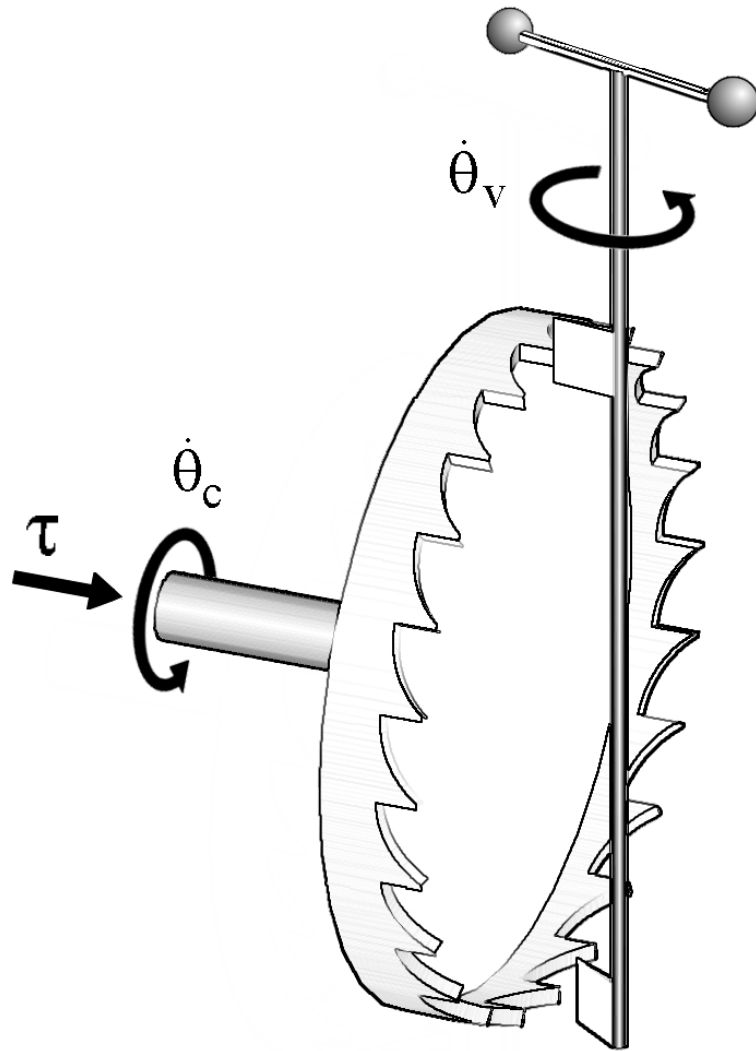
- How many limit cycles does a planar polynomial have?
 - Dulac's theorem 1923: Finite number for each system
 - Incorrect proof stood for 80 years—current status uncertain
 - For $n=2$, 4 are possible
 - For $n=3$, 11 are possible
 - For $n=5$, 24 are possible
 - No upper bound known for ANY n
- Besides the Riemann hypothesis, the most “elusive” of Hilbert's problems

$$\dot{x} = P(x, y)$$

$$\dot{y} = Q(x, y)$$

Hilbert's 16th P





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Ecospheres



Shrimp

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- System with constraint on a velocity..
 - But not a constraint on position



Turning radius is constrained

Multiple passes may be needed

The number of required passes increases as the turning radius decreases

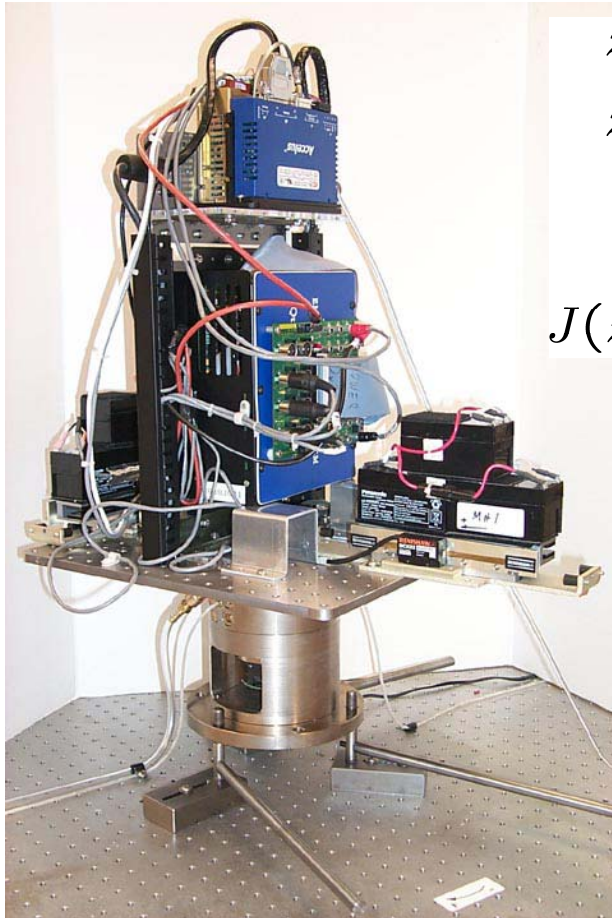
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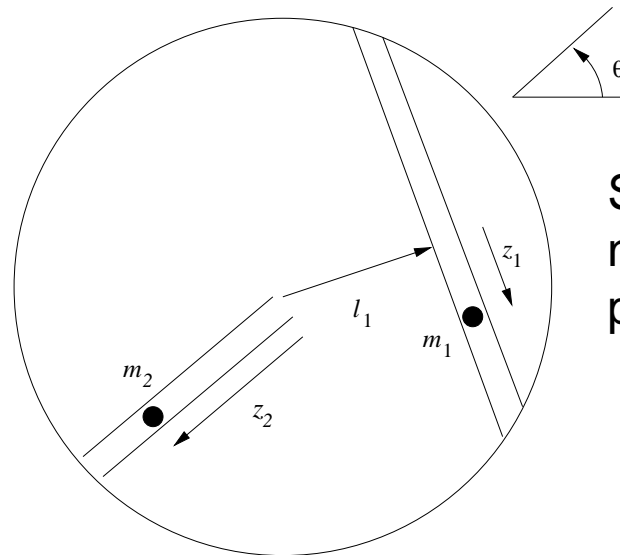
Shape Change Actuation

(Shen and McClamroch)

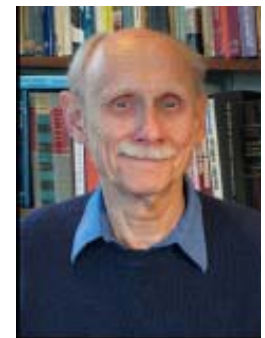
Angular momentum is conserved
.....but attitude is not constrained

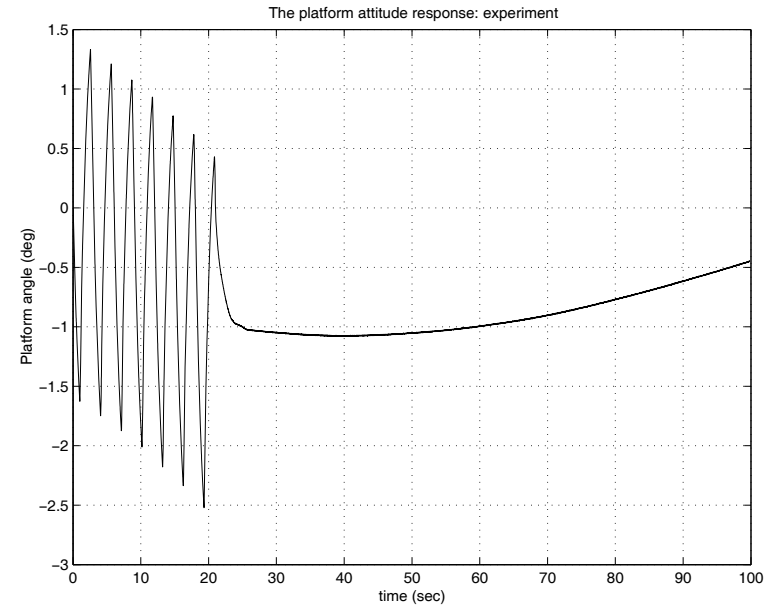
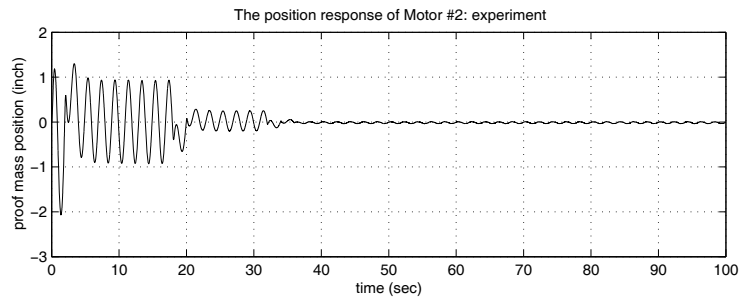
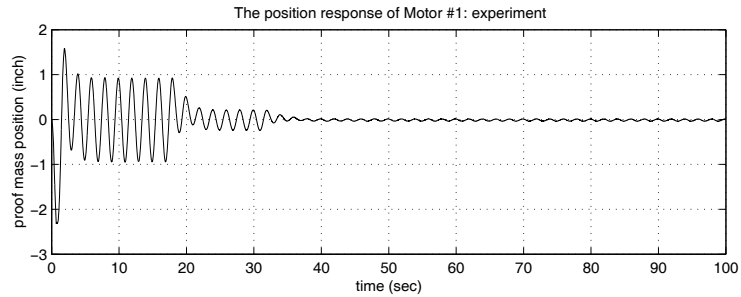


$$\begin{aligned} \dot{z}_1 &= v_1 \\ \dot{z}_2 &= v_2 \\ \dot{\theta} &= \frac{m_1 l_1}{J(z)} v_1 + \frac{m_2 l_2}{J(z)} v_2 \\ J(z) &= I + m_1(l_1^2 + z_1^2) + m_2(l_2^2 + z_2^2) \end{aligned}$$



Stroke constraints necessitate multiple passes

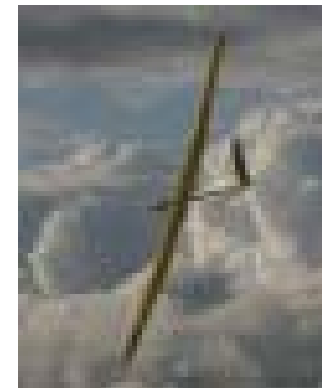
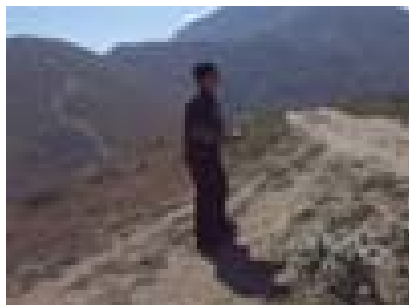




- How do birds stay aloft and cover huge distances with minimal energy?
- How can we keep an aircraft aloft indefinitely?

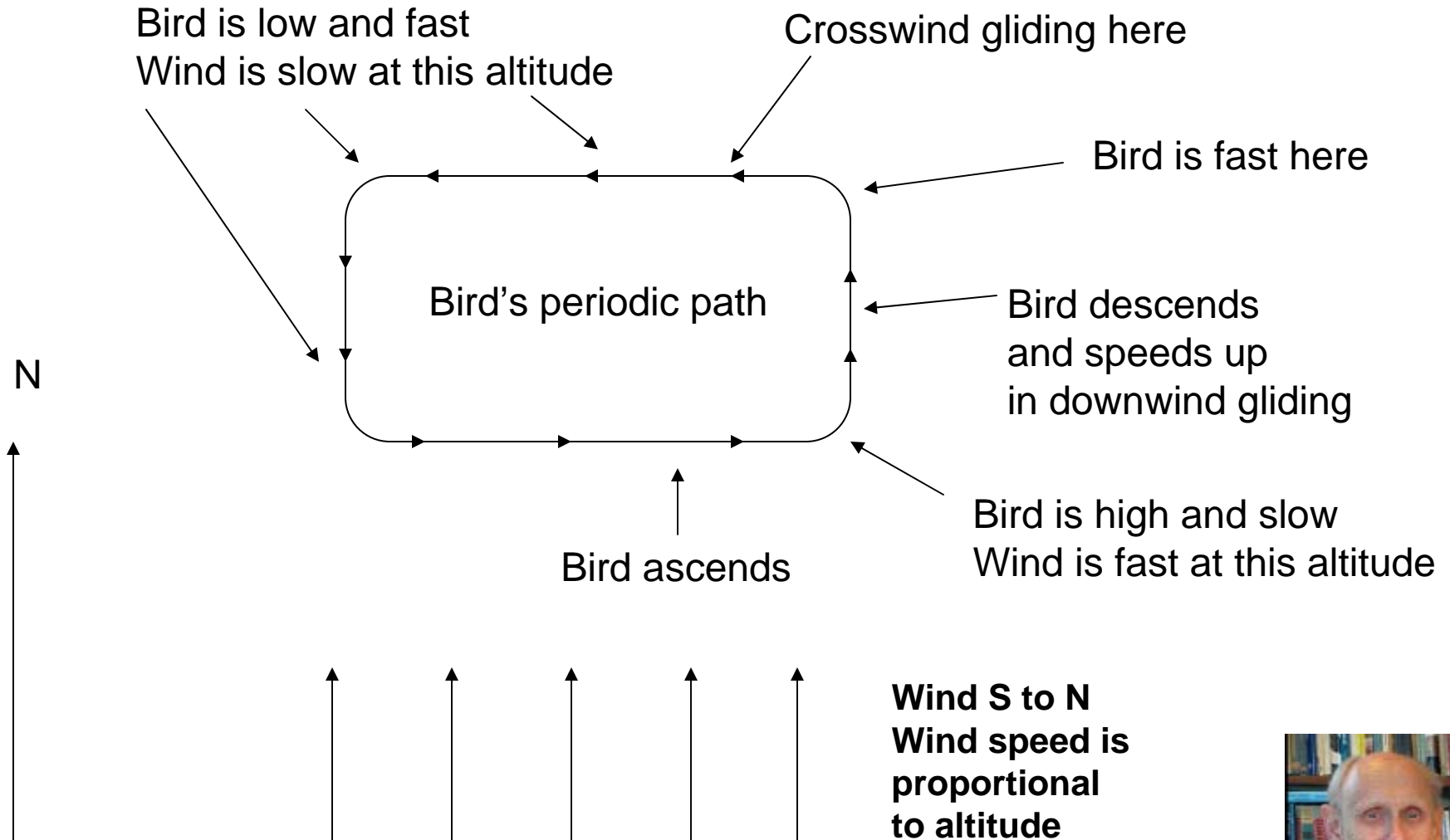
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- Harvesting energy from wind gradients—from special terrain
 - Not from vertical components/requires special conditions
 - Conjectured by Lord Rayleigh 1883 for albatrosses
 - Accomplished in 1974 by glider pilot:
 - “By repeating this manoeuvre he successfully maintained his height for around 20 minutes without the existence of ascending air...”
 - 392 mph RC glider record from 45 mph winds in 2009
 - UAV strategy to reduce fuel consumption
 - Zhao/Qi 2004
 - “All problem formulations are subject to UAV equations of motion, UAV operational constraints, proper initial conditions, and terminal conditions that enforce a periodic flight.”





Can we maintain arbitrary equilibria?

$$\dot{x} = Ax + Bu \Rightarrow 0 = A\bar{x} + B\bar{u}$$

- Why not?
 - Range of B is too small---need $m \geq n$
 - Unattainable equilibria
- Contradicts controllability??

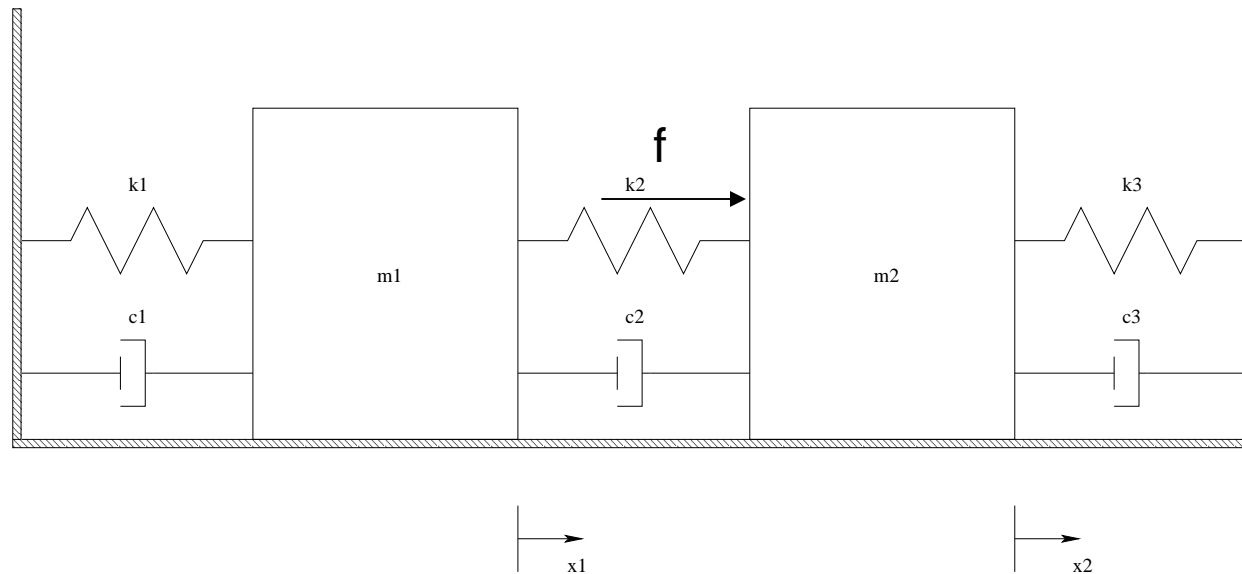
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- Vibration suppression
 - Bring motion to rest (origin) and stay there
 - E.g., vibrating membrane
- Assign shape
 - Bring motion to rest at desired equilibrium
 - Vibrating membrane with desired aperture shape



- 4-state structure is controllable with one force input
 - Can bring masses to arbitrary configuration at arbitrary time
 - Cannot stay there!
 - Desired equilibrium is unattainable
- Idea:
 - Reach, leave, and return quickly to desired “equilibrium”
- Do this periodically



Evolving Horizons in Systems and Control



- All airplanes are controllable in position and velocity
 - Can reach desired position with zero velocity
 - Not a good idea for most airplanes
- But most airplanes cannot hover
 - Not enough actuation
 - Works for helicopters
- Hummingbirds control periodically by flapping
 - Flapping induces small periodic motion
- Research problem: What is the best way to maintain operation near an unattainable equilibrium?
 - Loitering limited by actuation constraints
 - Barabanov “Non-assignable Equilibria,” Automatica, 2007



- Slow switching
 - Quasi steady state (QSS)
 - Convexify SSs to meet constraints
- Fast switching
 - Relaxed steady state (RSS)
 - Convexifies velocity set
- Slow switching between RSS's
 - Quasi-relaxed steady state (QRSS)
 - Convexify RSS's



The vehicle cruise problem is formulated as follows. The performance function

$$J(T(\cdot), V(\cdot), \tau) = V_{\text{avg}}(F_{\text{avg}})^{-1} \\ = \text{specific range} \quad (1)$$

depends on the thrust $T(\cdot)$ (measurable on $[0, \tau)$), the speed $V(\cdot)$, and the period $\tau > 0$ which satisfy the following constraints:

$$\dot{V} = -D(V) + T(t), \quad V(0) = V(\tau) \geq 0, \quad (2)$$

$$0 \leq T(t) \leq 1, \quad a.a. \ t \in [0, \tau], \quad (3)$$

$$F_{\text{avg}} = \frac{1}{\tau} \int_0^{\tau} F(T(t)) dt = \text{average fuel rate}, \quad (4)$$

$$V_{\text{avg}} = \frac{1}{\tau} \int_0^{\tau} V(t) dt = \text{average speed}, \quad (5)$$

$$V_{\text{avg}} \geq V_{\text{min}} \geq 0. \quad (6)$$

The condition $V(0) = V(\tau)$ assures that both $V(\cdot)$ and $T(\cdot)$ are periodic when the domain of these functions is extended to $(-\infty, +\infty)$ by

$$V(t+\tau) = V(t) \quad \text{and} \quad T(t+\tau) = T(t).$$

Gilbert, Automatica, 1976



- SS—equilibrium solution
- Relaxed steady state—fast switching between SS's
- QSS—slow switching between SS's
- QRSS—slow switching between RSS's

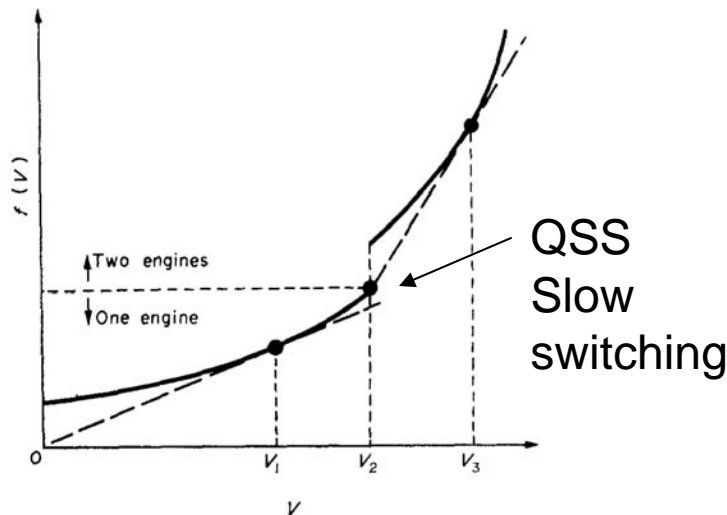


FIG. 3. Steady-state fuel rate for ship.

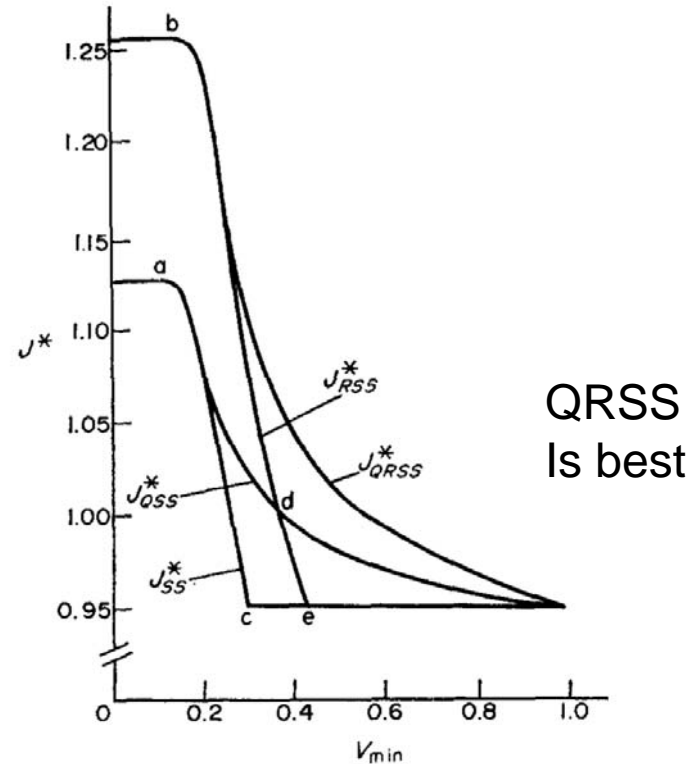
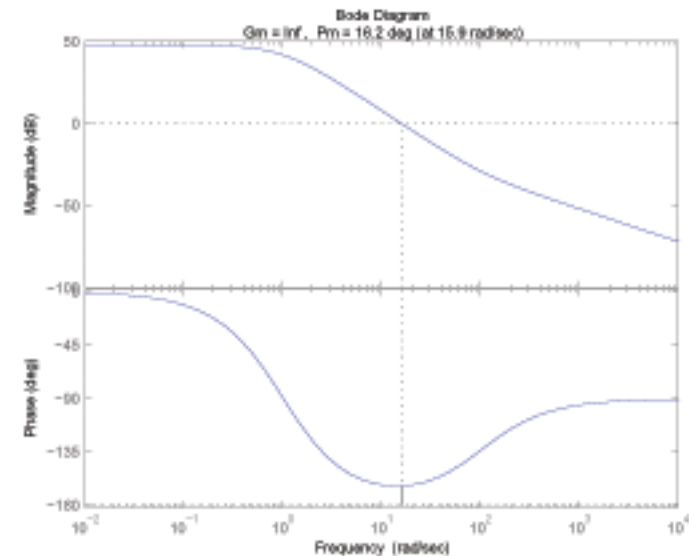


FIG. 2. Optimal cost functions for Example 1.



- Fundamental theorem of linear systems
 - Sinusoidal forcing of an AS LTI systems is eventually periodic
 - Gain and phase captured by Bode plot



- Periodic forcing is a special case of periodically time-varying dynamics
- When does a nonlinear system have the property “periodic-input/eventually-periodic-output”?
- Subharmonic, superharmonic, and nonperiodic solutions may exist
- Extremely complex problem

$$\dot{x} = f(x, u)$$

$$\dot{x} = f(x, \sin \omega t)$$

$$\dot{x} = f(x, t)$$



- Dynamics and cost $\dot{x} = f(x, u) \quad J = \frac{1}{\tau} \int_0^\tau g(x, u) dt$
- Find optimal steady state solution (\bar{x}, \bar{u})
- Linearize the cost and dynamics $\delta\dot{x} = A\delta x + B\delta u$
- Note $\delta J = 0$
- Evaluate $\delta^2 J$ with $\delta u = \sin \omega t$
- Find ω such that $\Pi(\omega) < 0$

$$\Pi(\omega) = G^*(\omega)L_{xx}G(\omega) + G^*(\omega)L_{xu} + L_{ux}G(\omega) + L_{uu}$$

– Pi Test----Guardabassi

- Then periodic control can locally do better than constant control



- Periodic control is necessitated by
 - Unassignable equilibria
 - Constraints
 - Nonconvex velocity set
- Periodic control can do dramatically better than constant control
 - Dynamic soaring
- Periodic control ensures sustainability
- Nature and humans have discovered these advantages and benefits



- Thank you, Elmer, for your constant guidance and wisdom, and for always setting the highest example of scholarship and integrity

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